

$$S = 2 \cdot \frac{M_g \cdot g}{Z_{av} \cdot R \cdot T_{av}} \cdot L \cdot \cos(\alpha) = 2 \cdot \frac{28.97 \cdot \gamma_g \cdot g}{Z_{av} \cdot R \cdot T_{av}} \cdot L \cdot \cos(\alpha)$$

$$S_q > S_c$$

$$L_q > L_c$$

TPR $q_{well} = C_T \left(\frac{p_{wf1}^2}{e^s} - p_{wh}^2 \right)^{0.5}$

eq
(1)

new unknown

2

IPR1 $q_{j1} = C_{p1} (p_a^2 - p_{wf1}^2)^{n_1}$

1

1

CPR $q_{j2} = C_c \left(\frac{p_{wf2}^2}{e^{s_0}} - p_{wf1}^2 \right)^{0.5}$

1

2

IPR2 $q_{j2} = C_{p2} (p_a^2 - p_{wf2}^2)^{n_2}$

1

0

mass
generation

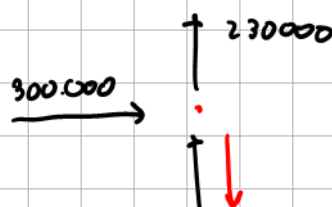
$$q_{well} = q_{j1} + q_{j2}$$

1	0
5	5

✓

$p_{wf1} = 150 \text{ bara}$

||
)



pwh [bara]	pwf1 [bara]	qwell [Sm3/d]	q1 [Sm3/d]	q2 [Sm3/d]
10	150.0	2.34E+05	3.23E+05	-8.94E+04

	pR [bara]	C [Sm ³ /bar ² n]	n
Layer 1	200	2010	0.52
Layer 2	250	1150	0.54

$$q_g = c(p_R^2 - p_{wf}^2)^n$$

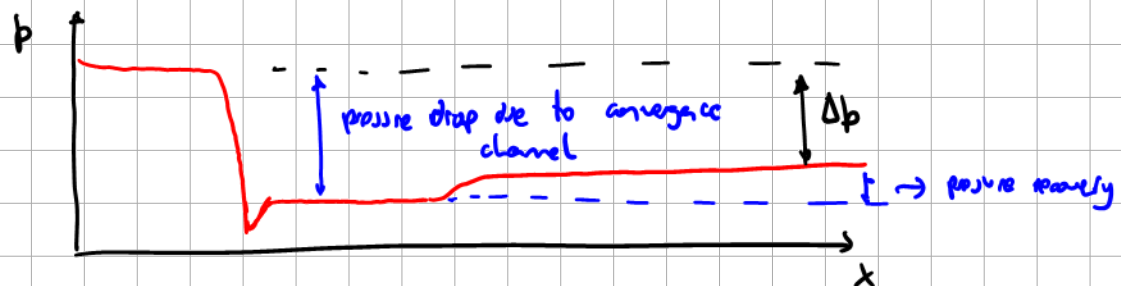
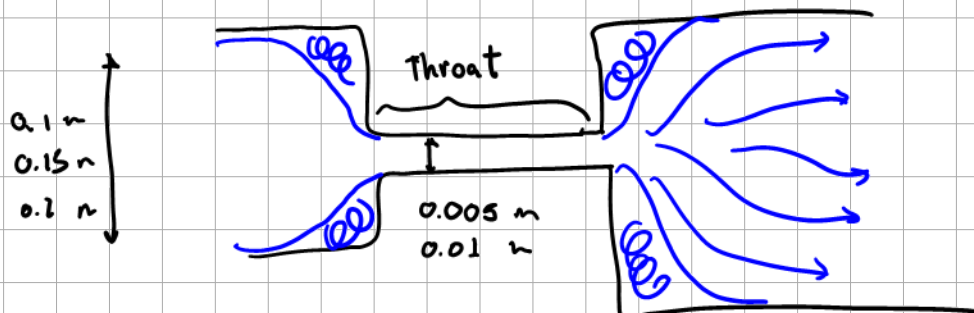
	Cf [Sm ³ /d/bar]	s
Tubing	2.09E+03	0.58
Casing between 1-2	4.00E+03	0.02

Tubing performance relationship (TPR)

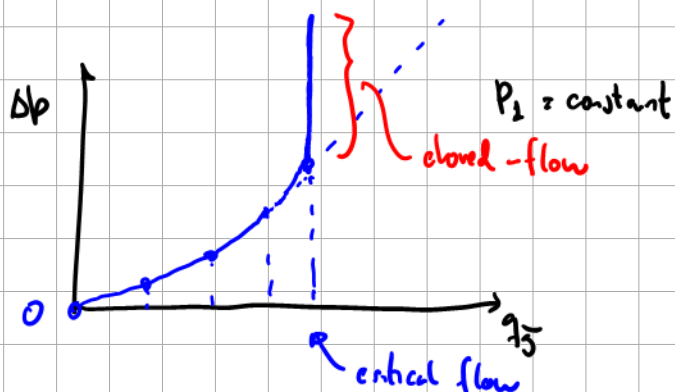
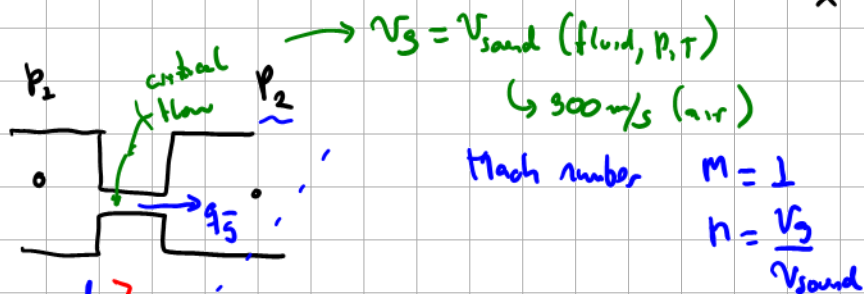
$$P_{wh} = P_2 = \left(\frac{p_1^2}{e^s} - \frac{q_g^2}{C_T^2} \right)^{0.5}$$

pwh [bara]	pwf1 [bara]	qwell [Sm ³ /d]	q1 [Sm ³ /d]	q2 [Sm ³ /d]	pwf2 (from TPR) [bara]	pwf2 (from IPR) [bara]	error [bar]
10	197.5	3.08E+05	7.21E+04	2.36E+05	208.1	208.1	0.0

Production cone



Dry gas



$$\frac{p_2^{out}}{p_3^{in}} = 0.5 \quad (0.4 - 0.6)$$



- flow performance $f(P_{in}, P_{out}, q_s, Q_{throat})$

- temperature performance

B. CHOKE EQUATIONS

UNDERSATURATED OIL FLOW

Based on a frictionless flow contraction from an upstream point 1 to a downstream point 2.

The single-phase Bernoulli equation for steady state frictionless flow along a streamline, neglecting elevation changes, is:

$$\frac{dp}{\rho} + V \cdot dV = 0 \quad \text{Eq. B-1}$$

Where:

p Pressure

ρ Density

V Velocity

$$P_{throat} \approx P_2 \quad \Delta P_{recovery} \approx 0$$

undersaturated oil

$$q_o = \frac{A_2 \cdot C_d}{B_{o,@2}} \cdot \sqrt{\frac{2 \cdot (p_2 - p_1)}{\rho \cdot (1 - \beta^4)}}$$

dry gas

$$q_g = \frac{p_1 \cdot T_{sc} \cdot A_2 \cdot C_d}{p_{sc}} \cdot \sqrt{2 \cdot \frac{R}{Z_1 \cdot T_1 \cdot M_w} \cdot \frac{k}{k-1} \cdot \left(y^{\frac{2}{k}} - y^{\frac{k+1}{k}} \right)}$$

$$\frac{P_2}{P_1} \rightarrow y_c = \left(\frac{2}{k+1} \right)^{\frac{k}{k-1}}$$

adiabatic exponent of gas
 $k = \frac{C_p}{C_v}$

$$k = 1.3$$

$$\frac{P_2}{P_1} > y_c$$

$$\frac{P_2}{P_1} < y_c$$

$$q_g = \frac{p_1 \cdot T_{sc} \cdot A_2 \cdot C_d}{p_{sc}} \cdot \sqrt{2 \cdot \frac{R}{Z_1 \cdot T_1 \cdot M_w} \cdot \frac{k}{k-1} \cdot \left(\frac{2}{e}^{\frac{2}{k}} - \frac{y}{c}^{\frac{k+1}{k}} \right)}$$